

AVIATION WEEK

MAR. 7, 1949

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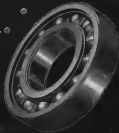
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controls cost
less than
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AERONAUTICAL CONTROLS

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such as landing gears, hydraulic struts, hydraulic actuators, gear boxes, transmissions, superchargers and variable speed drives, alternator drives and pressure regulator valves, Axelson is considered first choice by world leaders in aircraft manufacturing.

Axelson is currently producing its perimeters for the extensive production of the Douglas DC-4 triplane. Numerous Axelson experimental projects are under way, in design stage, production stage and in actual operating form. Axelson engineering maintains close contact with its customers to provide them efficient equipment, insuring economy with finest quality.



AXELSON

MANUFACTURING COMPANY

AIRCRAFT DIVISION

6140 South Bay Ave.
Los Angeles 11, Cal.

Vol. 58, No. 18

AVIATION WEEK

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The American Society of Mechanical Engineers (ASME) is a professional organization of engineers and scientists. It was founded in 1880 and is now the largest of its kind in the world. The society is composed of more than 100,000 members, including individual engineers, scientists, and students. The society's primary purpose is to advance the science and art of mechanical engineering. It does this through a variety of activities, including the publication of technical papers, the holding of conferences and exhibitions, and the provision of educational programs. The society also works to improve the standards of the engineering profession and to promote the use of mechanical engineering in industry and commerce. For more information about the society and its activities, contact the American Society of Mechanical Engineers, 345 Park Avenue, New York 17, N. Y.

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AVIATION WEEK, March 7, 1949

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Meanwhile, Sperry research and engineering development keep keeping ahead... improving old products, designing new ones.

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Skycouch Success

For American Airways' much-attested between New York and Puerto Rico is among links with controls flying the route.

In a last completed last year for CAB, FAA listed 57 authorized operation routes on the Puerto Rican route (beginning 1947 or the first part of 1948). But now, one of the controls reports, only about six regular operations are still in business, and perfectly all of these have been secured by CAB of flying too frequently an violation of the one scheduled exception.

Meanwhile, Eastern Air Lines, which last fly from New York to San Juan via Miami, indicates it will offer a permitted to operate mainly.

Subsidy Law

Enactment of legislation requiring operation of "error" from "subsidy" cost payments to airlines by Congress this session is a virtual certainty. The legislation, however, has been reported and will go into a bill by the House Commerce's recent release.

An Transport Act last year right could lengthen the operation, but such only will not at this time. Some still early-estimated savings have come around in the belief that flying program in the current that are "living off" the government would free them of the cost, even the very for certain all economic means, and be elsewhere for the industry.

At most, ATA is expected to offer only minimum assistance to operation.

Transport Flight

There is evidence of a major adjustment of interests in the transportation field.

An Transport Act has passed hands with the Air of American Airlines in the Transportation Act of America, launching an all-out campaign to secure the economic stability of the transportation industry and avoid all nationalization.

Now there is considerable speculation in route routes that National Independent Air Carrier, representing one scheduled air carrier, will join hands with the Federation of Airline Pilots headed by active vice-president Robert R. Young.

Joseph Boon, Washington repre-

NEWS SIDELIGHTS

Price Problem

President Truman's passing, concern on the high cost of military aircraft at the USAF Andrews Field Congressional demonstration has led investigation in the Pentagon.

Since then, several proposals reviewed with congressional have been called on the Pentagon cap and reduced on the need for economy in handling aircraft procurement funds and the taxpayer's dollar. Some observers believe that the cost of military aircraft has risen higher than required by prior rates in labor, material and increased complexity of design.

Which for increased emphasis on the need for a performance standard in the National Military Establishment that will detail its economy in such type of military effort other than the present system that made it impossible for the old Air Transport Command to compete the cost of its operation or get a clear idea of what any particular component of the aviation really costs the taxpayer.

initiative of the Federation, also agree with NIAC. A few years back, Young, who controls several seats in both chambers of the Congress, Congress studied out of AAR with a savings of millions—“business-controlled,” “non-competitive,” “non-interfering,” “government subsidized.”

Young's objectives in AAR follow the line of NIAC's objectives to ATA. He has made one step, however, toward his objective of building up a cash-investment net by purchase and control of the AAR trust. Joint Department has supported both Young's aim and NIAC's one.

Largier for Whitney

Washington observers are watching for significance of the shift of Tom Largier, into C. V. Whitney's job as Assistant Secretary of the Air Force.

Largier, a World War II fighter ace, former Air Force Air Corps pilot, and now a House member, after, has been active in Air Force affairs and recently has been backing of Floyd Collins and the Collins.

Whitney has been under fire lately in USAF for his handling of Air

Research matters. Whitney, a former Air Force Air Corps pilot, will become American Airways director, will become Undersecretary of Commerce when he will be expected to both Assistant Secretary John A. Loebe and CAA Administrator Dell R. Kestel. Coast Mission, Whitney's assistant in the Pentagon will move with him to the Commerce Dept.

Fat Lobby

Two of the organizations reporting the largest expenditures over the past year under the lobby registration provisions of the 1946 Congressional Reorganization Act are Transportation Association of America and Air of American Airlines.

TAA reported an expenditure of \$273,194 and AAR, \$126,679. TAA's legislative activities are directed by Clarence Lee, former chairman of the House Interstate and Foreign Commerce Committee, who raises a salary of \$15,000 a year, and Donald Conn, who earns \$15,000.

TAA recently announced that it had engaged Dr. John F. Friedman, former director of the professional staff of House Interstate, at a salary of \$10,000 a year for past year work.

Air Tramps

The "tramp" movement among of the air is evidenced by former CAB Chairman James M. Loebe, a gradually being developed on a world-wide scale by U. S. certified and noncertified airlines.

For American's C-46 cargo opens route to Latin America as well as a lead out of the newspaper advertisements of Steve Canyon.

Some other agencies, such as Alaska Airlines, Seawind & Western, Transamerica and Pan-Caribbean, which are based by CAB regulations from making common carrier passenger flights from the U. S. to other countries, have developed a flourishing business flying from foreign ports to foreign ports. Establishment of the new code of law has resulted in unprofitable contracts for the transportation into Alaska of New York, Alaska, Alaska (Alaska) and Alaska.

Trans-Atlantic contract flights in support of U. S. military commitments in Europe are expected to continue at least through this fiscal year. Development of President Truman's "global for itself" previously would open new business for tramp operators.



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AVIATION CALENDAR

Mar. 2-10—Schedule of Radio Electronics Conference, Hotel Commodore, New York City.

Mar. 10-12—Annual meeting, New York section, IAS, RAE, Section-101 Building, Radio City, 100 N. 4th St., New York City.

Mar. 10-12—Annual meeting of American Society of Test Engineers, Hotel William Penn, Pittsburgh.

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More horsepower per pound is delivered by the new Indian Scout line is achieved with 22 Needle Bearings in the transmission. Two Needle Bearings in the main shaft give increased stability, ensuring efficient seating with the standard oil chamber ground in behind them.



Manufacturing economy is secured by the easy installation of Needle Bearings. In rough, early-fabricated housings, Needle Bearings mean operating precision, too—minimum vibration for maintenance and longevity. Past experience indicates these all-steel bearings will serve the life of the motorcycle.

Lightweight design, high capacity, long service life and economical production can be yours with Torrington Needle Bearings. Our engineers will be glad to help you adapt these compact anti-friction units to the requirements of your equipment. Write us today. Torrington Company, Torrington, Conn., or South Bend 21, Ind. District offices and Distributors in principal areas.



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633—Ford; 634—General; 635—Ford; 636—General; 637—Ford; 638—General; 639—Ford; 640—General; 641—Ford; 642—General; 643—Ford; 644—General; 645—Ford; 646—General; 647—Ford; 648—General; 649—Ford; 650—General; 651—Ford; 652—General; 653—Ford; 654—General; 655—Ford; 656—General; 657—Ford; 658—General; 659—Ford; 660—General; 661—Ford; 662—General; 663—Ford; 664—General; 665—Ford; 666—General; 667—Ford; 668—General; 669—Ford; 670—General; 671—Ford; 672—General; 673—Ford; 674—General; 675—Ford; 676—General; 677—Ford; 678—General; 679—Ford; 680—General; 681—Ford; 682—General; 683—Ford; 684—General; 685—Ford; 686—General; 687—Ford; 688—General; 689—Ford; 690—General; 691—Ford; 692—General; 693—Ford; 694—General; 695—Ford; 696—General; 697—Ford; 698—General; 699—Ford; 700—General; 701—Ford; 702—General; 703—Ford; 704—General; 705—Ford; 706—General; 707—Ford; 708—General; 709—Ford; 710—General; 711—Ford; 712—General; 713—Ford; 714—General; 715—Ford; 716—General; 717—Ford; 718—General; 719—Ford; 720—General; 721—Ford; 722—General; 723—Ford; 724—General; 725—Ford; 726—General; 727—Ford; 728—General; 729—Ford; 730—General; 731—Ford; 732—General; 733—Ford; 734—General; 735—Ford; 736—General; 737—Ford; 738—General; 739—Ford; 740—General; 741—Ford; 742—General; 743—Ford; 744—General; 745—Ford; 746—General; 747—Ford; 748—General; 749—Ford; 750—General; 751—Ford; 752—General; 753—Ford; 754—General; 755—Ford; 756—General; 757—Ford; 758—General; 759—Ford; 760—General; 761—Ford; 762—General; 763—Ford; 764—General; 765—Ford; 766—General; 767—Ford; 768—General; 769—Ford; 770—General; 771—Ford; 772—General; 773—Ford; 774—General; 775—Ford; 776—General; 777—Ford; 778—General; 779—Ford; 780—General; 781—Ford; 782—General; 783—Ford; 784—General; 785—Ford; 786—General; 787—Ford; 788—General; 789—Ford; 790—General; 791—Ford; 792—General; 793—Ford; 794—General; 795—Ford; 796—General; 797—Ford; 798—General; 799—Ford; 800—General; 801—Ford; 802—General; 803—Ford; 804—General; 805—Ford; 806—General; 807—Ford; 808—General; 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1147—Ford; 1148—General; 1149—Ford; 1150—General; 1151—Ford; 1152—General; 1153—Ford; 1154—General; 1155—Ford; 1156—General; 1157—Ford; 1158—General; 1159—Ford; 1160—General; 1161—Ford; 1162—General; 1163—Ford; 1164—General; 1165—Ford; 1166—General; 1167—Ford; 1168—General; 1169—Ford; 1170—General; 1171—Ford; 1172—General; 1173—Ford; 1174—General; 1175—Ford; 1176—General; 1177—Ford; 1178—General; 1179—Ford; 1180—General; 1181—Ford; 1182—General; 1183—Ford; 1184—General; 1185—Ford; 1186—General; 1187—Ford; 1188—General; 1189—Ford; 1190—General; 1191—Ford; 1192—General; 1193—Ford; 1194—General; 1195—Ford; 1196—General; 1197—Ford; 1198—General; 1199—Ford; 1200—General; 1201—Ford; 1202—General; 1203—Ford; 1204—General; 1205—Ford; 1206—General; 1207—Ford; 1208—General; 1209—Ford; 1210—General; 1211—Ford; 1212—General; 1213—Ford; 1214—General; 1215—Ford; 1216—General; 1217—Ford; 1218—General; 1219—Ford; 1220—General; 1221—Ford; 1222—General; 1223—Ford; 1224—General; 1225—Ford; 1226—General; 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Convair Delta wing Model 76C.

USAF Reveals 3 New Supersonic Planes

Latest jet-rocket fighters show results in field of high speed flight research.

By Robert McLauren

Transition of supersonic flight research into combat aircraft is revealed in a trio of new U. S. Air Force supersonic interceptors. These three new planes are the Lockheed XF-90, Republic XF-45, and an experimental flying model used in the Convair XF-72 project.

Supersonic Progress—The three aircraft represent a design program aimed at solving the most perplexing problems in the supersonic speed program. All have two features in common: a wing designed to alleviate the wave drag of supersonic speed, and composite turbo-rocket power. Some borrowing from German wartime research is inevitable in the designs but substantial refinement is vital together with a wholly U. S. contribution—deserve high—is involved in the designs.

Lockheed's XF-90, most conventional of the three, features three swept wing two Westinghouse J40 turbojet engines and two T602 H₂ thrust rockets for short duration acceleration. With a

stout top speed, highly producible structure, and a drag-reducing-by-the-purpose equipment layout, it is believed to hold first preference in the procurement process. Prototype is nearing completion at Burbank.

Boeing-Taper-Republic's XF-45 is a radical departure from familiar high speed design and features severely tapered wing in which the tip chord is substantially greater than the root chord. This wing is designed to solve the perplexing problem of preventing tip stalling at low speed at the convex total swept wing. By shifting the lift distribution forward, Republic engineers led by chief engineer Alexander Korshak—before they have solved the problem of generating adverse control at nonstalling speeds, when it is needed most.

The increasing trade-off towards this wing plan has increased the problem of raising the main gear and egress tendency has been to cut this, large diameter wheels. Republic engineers switched to small, tandem wheels in such main gear alleviating the storage

problem while increasing the footprint area as well as necessary.

Problems of structure and fuel capacity accentuated the storage of the constructed gas contained near the wing tip, which accentuates the wing structure design problem.

Rockwell-McDonnell—Internal layout of the XF-72 is similar, generally, to the F-4 Thunderbolt fighter, with the intake duct bifurcated around the pilot and compressed forward of the General Electric J-47 turbojet engine developing nearly 5000 lb thrust with water injection. Alternatively, the engine may be an Allison J-35 which, with water injection, produces 5500 lb thrust. In addition, four rocket nozzles are mounted in a vertical arrangement at the extreme tail of the airplane, two above and below the turbojet intakes. These rockets can be used either for lift assistance or for highspeed acceleration at some speed.

XF-90 has a span of about 30 ft and is more than 47 ft long. It weighs 15,500 lb and has a maximum speed of Mach number 1.5. It is equipped with pressurized cabin and oxygen, a simplified cockpit layout and two 25-mm cannons, mounted in the nose. Upon completion of two tests at Republic's Farmingdale, L. I. plant, it will be



Side view of Republic XF-45.



loaded aboard an Air Force transport and flown to Marine Air Force Base, Camp Lejeune, North Carolina.

Delta Wing Model—Although the Delta wing Convair Model 7602 is a second flying prototype, its two half seconds USAF and Convair design thinking on the XF-72 jet-rocket fighter, further development of which has been controlled. The original design, with an estimated top speed of 1165 mph, as Mach No. 1.75, was to have been powered by a ramjet engine with a series of ducted rocket nozzles for lift-off and acceleration to supersonic speed. This rocket sustainer system use of a jet engine tube, similar to that used on the Convair-Letter ex-haust system, to accelerate an auto a highspeed jet that produces additional thrust over that of the rocket alone.

Shock Delta—In addition to the considerable development problems posed by the radical powerplants, the problem of

the Delta wing was not thoroughly understood and it was decided to build and test a flying model of the design to obtain stability and control data in the Delta wing configuration. The resulting research airplane is the Convair Model 7602, which has made more than 18 successful test flights.

Model 7602 is powered by an Allison J-43 turbojet flow turbojet engine developing 5200 lb thrust with water injection. Since it is to be used only for low speed (450-500 mph) stability tests, rocket nozzles are not fitted.

Altitude Construction—The Delta wing is a logical conclusion of the effect of combined sweepback and low aspect ratio, which must be used in combination as the sweep angle is increased. It also reduces the structural problems posed by large sweep angles by generating a straight, lateral edge across the trailing edge. This now provides the stiffness required to prevent wing flexion from changing the lead line en-

efficient and thus compromising the efficiency of the wing.

By sweeping back the wing to 60 deg., the critical Mach number is raised well into the supersonic realm and the drag rise at supersonic speed is delayed and alleviated. These highspeed advantages are accompanied by serious trade-off problems that are now being explored by the Model 7602. Although the Delta wing configuration is not subject to the tip stall phenomenon of the simple swept wing, it presents severe low-speed stability difficulties. One of these is an extremely high stalling angle that may require the pilot to nose the airplane up as high as 30 deg. before touchdown in order to obtain maximum lift and, therefore, maximum landing speed. The Model 7602 is provided with a heavy tailfin to prevent damage to the jet intakes during such high angle landings.

Tailfin Problem—Location of the control surfaces along the trailing edge



penders the use of flaps with a full-flap section of elevator left. The two trailing edge surfaces act in elevation when moved in unison and as ailerons when operated differentially. The Delta wing layout thus takes on many of the problems of the tailless airplane, with the exception of directional stability, which is provided by a large, swept fin and rudder.

The Model 7002 is of all-metal construction with a single rotor an inlet for the turbojet engine. The duct is bifurcated around the rotor and re-enters at the inlet to the engine nacelle. All fuel is stored in the fuselage, which contains the 7002 no. 3000 turbojet engine of about 14,000 hp. Extremely close frame spacing has been used as the backbone to provide stiffness.

► **Electric** Bell-Pfister's test is expected to show folding wings and rotor on hinges placed aft. Wing structure features a front spar swept back with nose to meet the leading edge and more like parallel to the trailing edge. The control surfaces are fabric covered, making any external bow balloons on the lower side of the elevator and the port side of the tailfin.

Longitudinal close struts are used on the leading spar to absorb the high vertical impact loads of landing. Nose wheel struts forward into clamshell doors and the main gear folds outboard into the wing, where it is covered by large landing panels upon touchdown. Model 7002 is completely instrumented as a research airplane with a multi-channel telemetry system transmitted to the test room at the intake duct. Airspeed land is located atop the fin out of reach of the aircraft's shock wave that would be created by supersonic flight of the craft.

As Bell-Pfister has no immediate plans for further development of the Delta wing configuration pending outcome of tests as the 7002.

Hughes Flying Boat Damage Discovered

Recent discovery of an unknown origin has been discovered in the last assembly of Hughes Aircraft Co.'s giant flying boat. Edward Hughes told *Aviation Week* that the unknown discovery of an elevator, rudder and fin in one he had been extensively damaged to a degree which would have been critical and could have caused a crash had it gone unobserved until completion of flight tests.

Extent of the hidden damage, found several weeks ago by workers mending elevator test equipment, was disclosed but work Hughes has discussed the matter with the Federal Bureau of In-

vestigation, but refused to say whether airplane was suspected.

Five tests of the flying boat may be started this month as soon as it has been equipped with new invisible-pitch reversible standard propellers.

During the year the plane has been in stock since its first flight, Hughes reportedly has spent approximately \$2 million on modifications. The Civil Aeronautics Administration office at Los Angeles indicates that up to 10 changes have been made.

Major changes include installation of a "swing tab" dynamic control system in addition to the boat's original hydraulic control system, reduction of eight degrees in lift, replacement of pneumatic engine controls by electric controls, general revision of instrument panels and engine's controls, and installation of special frame and fin to the deck to protect the pilot against smoke in event of fire under the plane during flight.

Schildhauser Resigns

Capt. C. H. "Doc" Schildhauser, USAF (Ret.), has resigned from the Civil Aeronautics Administration, resigning from the Navy's flying boat design branch during World War II and now serving in the sales department of Boeing at the time of his resignation.

Skycoach Expansion

Capital Airlines is seeking CAA permission to expand its highly experimental Skycoach operation. The company, which emerged from low-cost New York-Pittsburgh-Chicago flight last November (see *Aviation Week*), now plans to provide similar service between New York, and the Twin Cities via Pittsburgh, Cleveland, Detroit and Milwaukee starting May 24, and between Washington and Chicago via Pittsburgh, Cleveland and Detroit starting Apr. 3. Like the original Skycoach operation, Capital's new Skycoach flights will have three terminals between midnight and 3 a.m., fifty-one passenger DC-4s will be used, and fares will be about 4 cents a mile.

Northwest Airlines has asked CAA authority to open Skycoach service between New York and Seattle via Minneapolis and other points on Mar. 24 with 50-passenger DC-4s (*Aviation Week*, Feb. 21). Thus Capital and NWA may not be competing for two-class business on the New York-Twin Cities run.

Military Deliveries

Rose in 1948

Military aircraft production rose steadily during 1948 according to delivery figures of the U.S. Air Force and Navy.

Total military production increased from 11,998,000 lbs. of airplane weight during 1949 to 12,143,000 lbs. in 1948. Airplane deliveries to the service rose in a steady climb during the year to a peak of 1,276,600 lbs. for December, 1948. Three deliveries reflected only slightly the increased aircraft production, which was the result of the 50th Congress in the spring of 1948.

Rate of the deliveries under that program will be reflected in deliveries for the last half of 1948 and much a peak, in 1950.

Following are the 1948 monthly deliveries of military aircraft to both USAF and Navy expressed as thousands:

Jan	1,251,000
Feb	1,464,000
March	1,567,000
April	1,527,000
May	1,455,000
June	2,037,000
July	2,561,000
Aug.	2,668,000
Sept.	1,677,000
Oct.	1,735,000
Nov.	2,695,000
Dec.	1,246,000
Total (y.)	25,143,000

Andrew Willgoos Dies, P-W Chief Engineer

Andrew V. D. Willgoos, 66, chief engineer, Pratt & Whitney Aircraft, died suddenly of a heart which was showing signs of an earlier heart attack, heart.

The chief engine designer met one of the original engineers of Pratt & Whitney, joining its founding members in 1925 in the design of the first "Wasp" engine. He continued as chief research and design engineer through out his long period of service.

Although noted chief engineer, Willgoos devoted all of his time to "designing" new and more powerful propellers as director of a small experimental design group. He was personally responsible for the wartime Double Wasp and the Wasp Major, but from 1946 he worked on gas turbine products. He and his brother Thomas, who became P-W chief of production during the war, were one of the company's research staff as master mechanics, and preserved their interest in precision machine work throughout their careers.



New Helicopter Sets Speed Mark

Seven-place Sikorski XHJP-1, designed for use by Navy on ships, achieves unofficial record of 131 mph.

A new unofficial helicopter speed record of 131 mph. has been set by the Sikorski XHJP-1, a four rotor aircraft designed for the Navy.

The XHJP-1 was set in motion at a calibrated runway at Patuxent, Md., plant, official world record is held by the British Fairey Gyron at 128.6 mph. American record is 114 mph set by a Sikorski S-51.

► **Four** Sikorski-Labeled Patuxent model in competition with the Sikorski HJ-1 for acceptance as the standard Navy helicopter for first use. Any plans to use this type for ship to shore and ship to ship communications, rescue work, observation, and personnel transport. It must be suitable for operations of several engines, landing and hovering. Navy currently estimates it will need 150 of this type helicopters.

The XHJP-1 is of all metal construction to induce vibration and fastest landing maneuvering rates fast and slow and a large vertical fin for directional stability. It carries a crew of two and five passengers and is equipped with internal main rotor and hydraulic lift for landing from directly into the water.

► **Four** Sikorski-Overall dimensions are designed to fit any carrier elevator without folding rotor blades and any carrier elevator with blades folded. Dimensions with blades folded: 71 ft. 7 inches long, 4 ft. 4 inches wide and 12 ft. 6 inches high, with blades extended it is 59 ft. 1 inch long and 10 ft. 4 inches wide.

It is powered by a Continental R-757-14 radial engine rated at 525 hp for take-off engine not cooled through an intake in the vertical fin with auxiliary engine below the fuselage. Single engine drives both rotors. The

main rotor blades are made of steel tubing with plywood covering. Tensioned all metal blades will be added to later models with an expected increase in top speed to 147 mph.

► **Crew** at 114 mph—Forward engine speed is 114 mph, at 75 percent of power. The XHJP-1 has four sets of four main rotors at 125 mph. It has a 1800 rpm rate of climb and has flown up to 16,000 ft.

Crew seats are mounted side by side at the rear. Design was begun in late 1945 with the first model flying in March, 1946. Two models are currently at Naval Air Station, Patuxent, Md., where they have been undergoing evaluation tests with the Sikorski HJ-1.

► **Many** maintenance—The XHJP-1 features built-in steps for easy access to all parts of the fuselage and blades and convenient access panels for simplified maintenance. The power panels can be changed in a moment. Configuration of the fuselage was designed for simple manufacturing requiring only single piece forming for most of the external parts.

The XHJP-1 is also a flying water model of the XH-16, a U.S. Air Force experimental transport helicopter with detachable landing gear.

Election Void
Results of a recent day-long election election held by the CIO United Automobile Workers of America, was declared void. Workers at four plants at North American Aviation last August have been set aside by the National Labor Relations Board.

Under the election, conducted by mail, was managed by the regional NLRB director. Many employees did not get ballots, and little more than half the workers voted.

The vote was declared void by the union's national board, and the Taft-Hartley law, a majority of the eligible employees have to appear before the union may begin for a union security clause in its contract. The vote was 614 to 249 for a union shop, but over 15,532 were eligible, the 6314 favorable votes was 1/3 short of the necessary number.

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Goodyear Plant
Goodyear Tire and Rubber Co. has announced plans to open its plant at Louisville, Ky., for the manufacture of tire casings for the prototype of the Navy ship to be built for the Navy.

Operation will employ about 100 workers, and is expected to be a loss-maker of an expanded program. A. J. Thomas, Goodyear company president, reported.



General view of the XHJP-1

BRIEFING FOR DEALERS & DISTRIBUTORS

FARM AND BUSINESS SURVEY—A survey currently being made by the Personal Aircraft Council of Aircraft Industries Assoc. among 2000 farm plane owners and 2000 business plane owners is expected to give the land-based manufacturers some additional insight into customer requirements. And a special question addressed to the owner is likely to provide some additional evidence in support of GE rotational flight training. The question seeks to determine how many employers in the group addressed consider the ability to fly a plane as important qualification for prospective employees.

PURDUE SHORT COURSE—Next in the list of short courses in aerial spraying and spraying have been conducted at various state schools in the recent months, preparing for greater volume of work this spring and summer, is the Purdue University two-day course May 19-21 at Lafayette, Ind.

Subjects: insect, mold and disease problems, chemicals, uses and limitations, aerial equipment, legal aspects, CAA and state airframes, use of the airplane in agricultural production, and a focus on operational problems of dusting and spraying by aircraft.

EXCLUSIVE GASOLINE FRANCHISE—CAA is adding aviation industry members on a proposed revision of regulations governing sale and delivery of gasoline and oil at airports participating in the Federal Airport Aid program. The new circulation follows criticism of the existing Part 350 of Administrator's regulations, which allows airport operators to pump or contract an exclusive right to sell aviation gasoline and oil on their airports but allows to all aircraft users on the airport the right to purchase gasoline and oil of the field to be delivered for their own use.

It is pointed however, that the airport operator may charge a reasonable fee representing services in connection with delivery of gasoline and oil. Views of airport operators, fuel tank operators, airlines, state, county and municipal associations and oil companies are asked in the consultation.

PIPER LANDING LIGHTS—Aircraft Accessories Co., which handles installations of optional extra equipment at Lock Haven for Piper airplanes is making available a CAA approved landing light installation for the PA 18 Family Cruiser and PA 12 Super Cruiser. It will be available as a new Family Cruiser before delivery at list price of \$500, or a lot can be purchased for field installation at \$36 list.

Installation fits into landing edge of fuselage of propeller drive, and includes two usual horn units—two directed for timing and one for landing, of 180-90° dimensions each.

BENDIX RADIO SALES UP—Sales of Bendix personal plane radios for January, 1969, were 50 percent better than in the same month in 1968. J. W. Coble, aviation sales marketing manager for the Baltimore Bendix division, reports Aviation Week. Coble said 75 percent of the radios were going into airplanes a year old or more.

He attributes the sales increase, which defies the down trend in new plane sales, to the fact that owners are coming to get greater safety from their planes and use the need for radios. Coble also reports that a steady portion of Bendix business is replacement for other makes of radios.

562 GOOD AIRPORTS—Continents of "Good Airport Operating Facilities" were awarded to 562 airports in 29 states in the 1968 NAA report period. Oakley, California, with the largest number of planes and pilots, qualified only one airport. Leading state was Indiana, with 33 airports. Second was Nebraska with 72 airports. Oklahoma ranked third with 56.

National Aeronautic Association, American Assoc. of Airport Executives and Airport Operators Councils cooperated with NAA in distributing the judging sheets to a total of 6000 airports. Fields qualifying for the certificate were required to comply fully with a specified list of required good practices, and to meet at least full requirements in a second list of desirable but not essential practices.

—ALEXANDER MURPHY

it is being conducted in addition to the Bendix line's regular engine service operations. Cing Paul is Jacksonville's Manager, Airport No. 1, a private Navy auxiliary field, where Bendix installed its own facilities to take advantage of the paved runways adjacent from the Navy.

Bendix has a small administrative building, restaurant, electronic administrative office, a large 100x105 ft. hangar and a new hangar with 50 7-ft-high stalls. Here, in addition to the engine program, Bendix carries on its regular airport service, gas and oil, shop, parts, supplies, and civilian flight training and charter service, following the routine of the ordinary fixed base operator.

Iowa Clinic

More than 200 maintenance flight operators attended the aerial spraying and dusting clinic held in conjunction with the Iowa Flight Operators Association in Des Moines recently.

Dr. Harold Goodenough, Iowa State College entomologist, stressed that with proper timing the men have could be controlled with aerial spraying and pointed out that the ability of the airplane to cover large territories rapidly was important in attacking this pest. Spraying in the midwestern area is needed principally for corn borer, soybean and weed control.

Harry Lind, Iowa agricultural secretary, urged that pilots engaged in spraying and dusting be required to pass a special examination on use of chemicals and their effects on various crops. Lind's statement was issued to correct legislation and self-regulation by the operators.

Earl J. Howard, Ames, was elected president of the Iowa operators, with J. A. Sirdy, Charles, as vice president, and Virgil Cline, Shenandoah, secretary.

Clinic was sponsored jointly by the operators, the Iowa Association Commission and Iowa State College.

More Farm Flying

Rapid growth of agricultural aviation activities, including spraying, dusting and seeding, is seen in NATA analysis of records showing 779 flight operations in the U. S. 8-11 acreage of approximately 16 in a state now engaged in these activities.

California leads the state with 96 operations, Texas is second with 62, and Florida third with 58. Other leading states: Kansas, 42; Iowa, 35; Illinois, 31; Arkansas, 29; Colorado, 28; Washington and Nebraska, 26 each.

The analysis was made on a basis of CAA and Department of Agriculture figures, in connection with the operators' development of an agricultural activities database of NATA.

Cold or fuelled spark points won't stop heater or de-icer ignition when you use General Electric aircraft transformers. These compact, 100-cycle units provide an unfailing high-voltage arc.

Now improved shielding prevents radio interference over wave bands between 10 kc and 300 mc. All ratings are corrected for power factor.

GE aircraft transformers are available for general delivery for both single- and two-ignition, with 70 Vu and 100 Vu capacity. Net weights range from 6½ to 33 lbs. For details on the complete line of General Electric aircraft transformers, contact your nearest G-E sales office, or write for Bulletin GEA-5166. Appearance Dept., General Electric Company, Schenectady, N. Y.

GENERAL ELECTRIC

Transformers for Aircraft

- Among General Electric aircraft transformers are the following:
- Ballasts for fluorescent lighting
 - Ignition transformers for carburetors, glow discharges
 - Ignition transformers for jet engines
 - Phase-shifting transformers
 - Ground-reference transformers
 - Ground-reference "through" offset transformers

- All General Electric aircraft transformers are designed to:
- Give dependable operation up to 100,000 hours
 - Operate in ambient temperatures from -100 to 250 °F
 - Maintain proper frequency variation to power supply
 - Withstand vibration and shock
 - Keep weight and size to a minimum

brut, propeller, accessories and code
torn and faded.

► **Felony Breakdown**—A portion of the data furnished by CALL is given in the accompanying extensive tabulation. A brief recapitulation of the felony gives these striking divisions:

Port System	920
Engine Structure	99
Lubricating System	47
Propeller Assembly	39
Ignition System	18
Control System	12
Miscellaneous	9
Engine Accessories	

Total Determined	1,037
Total Undetermined	278

Total	1,939
-------	-------

These figures show that 4 of the recorded accidents, and 2 of three of known cause, were brought on by human failure of the fuel system—obviously the most important part of the powerplant to be assessed.

Of the remaining $\frac{1}{2}$ of the detector powerplant inertia not associated with the ball system, about $\frac{1}{4}$ was judged of the engine structure, a little less than $\frac{1}{4}$ was judged of the lubricating system, and the remainder divided among the other parts of the powerplant.

★ **Suggested Solutions** — Most likely remedies for these failures, as the opinions of the writer, are grouped in the following four classifications (most of the feared findings could, of course, have been avoided if the planes had been equipped with two engines and were capable of sustaining a takeoff or flying away from a failed landing with one engine stopped).

A Duplication of system (such as two copies, fuel or lubrication system).

3 Change in basic methods (in fact sufficient to fuel injection systems, or characterisation of steam giving two 100)

C. Use of fuel system giving an available warning when fuel supply is left, and accurate indication of fuel quantity from that time on.

And where none of these reasons is applicable, apparently the only one left is:

D. Improved detail design, construction or maintenance.

Obviously, D will help study all types of mechanical difficulty, but it is listed here only for cases in which conditions A, B, or C are not available.

In the legislation, the recommended remedies are designated by these letters for the individual types of claims based

► **Engine Structure**—Considering the individual states in the listing, we find that unless a suitable two-engine arrangement is used, all of the power-

[illegible]

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1994, CA/1994. A new, double Mustang all-terrain chain saw with and four Engines without mounted and discharged under way. Thicker Plexiglas windshield reduce noise and new windows filter airborne dust. Low noise level in Rye, New York '94 was a new standard in cold. (See comment.)



PROVEN MECHANICAL EQUIPMENT. Exclusive dual fuel system with independently operated engine-driven fuel pump and standby electric pump ensures new *Maxim* with most sensitive six buoys for dependable fuel supply. New VHP engine is now standard equipment. Best value comparison—check against the rest.



IMPROVED CONTROL AND INSTRUMENT PANELS and more instruments... Manifold pressure, rise-of-charge indicator, outside temperature and dimmed fuel gauge now included. Panel lights can be dimmed. Customizable variable resistance fly across for peak identification. New low-cost



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place structural failures must be taken care of by remedy D—repeated detail design, construction or maintenance. Repeated detail must be avoided not for each individual design of engine, and reliable operation depends as well on good maintenance.

Some structural failures will always be with us, but conditions are very much better with present day engines than they were two years ago, and further improvement can be expected in the normal course of development.

Improvement can be expected, of course, by critical analysis of the accident records and by continuous, ardent effort toward betterment of details by engine manufacturers. It is probably false to expect early in the maintenance development of a single engine model over a period of years.

► **Fuel System**—Almost all of fuel system failures would have been learned by improved detail design, construction or maintenance, but could be avoided by other means as well. Actually, the only ones which could be reduced by improved detail design alone are those failures resulting from an engine lock, or an over-temperature or major vibration when fuel supply was low.

With the exception of these items, if appears that practically all forced landings because of failures under the headings "fuel leakage" and "interruption of fuel flow" could be eliminated by remedy A—duplication of the fuel system.

Probably, engine failures could be taken care of by remedy B, elimination of the primer. If use of a carburetor is to be continued, possibly an automatic fire acceleration pump may be substituted for the primer.

If the fuel injection system is used, possibly the injection can be made positive enough even at engine cranking speeds so that a primer will be unnecessary.

► **Carburetor Failure**—Under the heading "carburetor," which lists a total of 479 accidents, 462, or 92 percent, of the failures are ascribed to "inadequate carburetor design" and "inadequate application of heat."

A carburetor incorporating automatic heat control would help this situation. It appears, however, that possibly all of these failures could have been prevented by remedy B, eliminating the carburetor and substituting a fuel injection system.

It now seems to be recognized that possibly carburetor fuel injection systems are free from the carburetor icing hazard. If this is true, then only two of the main engine controls (throttle and fuel control) in conjunction with the carburetor and its affecting the pilot's pilot of the serious

Name of Occurrence	Primary Incidents				Secondary Incidents			
	Number	Percentage	Number	Percentage	Number	Percentage	Number	Percentage
Other Section Failures								
Unaccounted loss in operating	5	0.2	1	0.1	1	0.1	1	0.1
Unaccounted loss in engine installation	1	0.1	1	0.1	1	0.1	1	0.1
Unaccounted loss in engine installation	1	0.1	1	0.1	1	0.1	1	0.1
Subtotal	7	0.3	3	0.1	3	0.1	3	0.1
PROPPELLER, WING/ENGINE								
Engine/propeller assembly - outboard	4	0.2	1	0.1	1	0.1	1	0.1
Propeller failed in flight	0	0.0	0	0.0	0	0.0	0	0.0
Engine rotor off in flight	0	0.0	0	0.0	0	0.0	0	0.0
Flow control failure	0	0.0	0	0.0	0	0.0	0	0.0
Water intake	0	0.0	0	0.0	0	0.0	0	0.0
Protein cylinder assembly thrust bearing	1	0.1	1	0.1	1	0.1	1	0.1
Inlet	1	0.1	1	0.1	1	0.1	1	0.1
Push-pull control assembly	1	0.1	1	0.1	1	0.1	1	0.1
Fuel, outboard	0	0.0	0	0.0	0	0.0	0	0.0
Subtotal	10	0.5	3	0.1	3	0.1	3	0.1
LEGISLATION SYSTEM								
On power failed, outboard	0	0.0	0	0.0	0	0.0	0	0.0
On system stopped	0	0.0	0	0.0	0	0.0	0	0.0
Clipped in flight	0	0.0	0	0.0	0	0.0	0	0.0
Loss of engine	0	0.0	0	0.0	0	0.0	0	0.0
Low oil pressure	0	0.0	0	0.0	0	0.0	0	0.0
On system fuel failed	0	0.0	0	0.0	0	0.0	0	0.0
Fuel, outboard	1	0.1	1	0.1	1	0.1	1	0.1
Circuit in flight	0	0.0	0	0.0	0	0.0	0	0.0
Filter not reset off in flight	0	0.0	0	0.0	0	0.0	0	0.0
Engine stop, engine not in flight	0	0.0	0	0.0	0	0.0	0	0.0
Power loss	0	0.0	0	0.0	0	0.0	0	0.0
Fuel stop engine	0	0.0	0	0.0	0	0.0	0	0.0
Airflow failure	0	0.0	0	0.0	0	0.0	0	0.0
On engine fuel feeding thermal cut	1	0.1	1	0.1	1	0.1	1	0.1
Fuel, outboard	0	0.0	0	0.0	0	0.0	0	0.0
Engine stop	0	0.0	0	0.0	0	0.0	0	0.0
Subtotal	47	2.4	1	0.1	1	0.1	1	0.1
ENGINE ACCESSORIES								
Start normal starter motor	1	0.1	1	0.1	1	0.1	1	0.1
Starter motor shut off	1	0.1	1	0.1	1	0.1	1	0.1
Starter motor shall ground off	0	0.0	0	0.0	0	0.0	0	0.0
Subtotal	2	0.1	2	0.1	2	0.1	2	0.1
CONTROL SYSTEM								
Throttle control valve actuator	7	0.3	7	0.3	7	0.3	7	0.3
Throttle control valve linkage	0	0.0	0	0.0	0	0.0	0	0.0
Throttle control valve linkage	1	0.1	1	0.1	1	0.1	1	0.1
Throttle cable linkage failure	1	0.1	1	0.1	1	0.1	1	0.1
Throttle cable linkage disconnected	1	0.1	1	0.1	1	0.1	1	0.1
Subtotal	10	0.5	10	0.5	10	0.5	10	0.5
METRIC/ANALYSIS								
Engine outboard	0	0.0	0	0.0	0	0.0	0	0.0
Engine outboard	1	0.1	1	0.1	1	0.1	1	0.1
Outboard engine in flight	1	0.1	1	0.1	1	0.1	1	0.1
Subtotal	2	0.1	2	0.1	2	0.1	2	0.1
TOTAL								
100	5.0	25.0	100	5.0	100	5.0	100	5.0



Welding and Brazing

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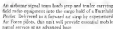
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AVIATION WEEK, March 7, 1943

Progressive additions to basic unit present major engineering problems.

By Robert McLaren

When the first news of the turbojet engine reached the layman, he was told with great acclaim that this astounding new engine had "only one moving part," the rotor, carrying the turbine and compressor. At the time, even heads nodded at this simplification and said, "Yes, but wait until the engineers get hold of it!"

In the ensuing five years the company has, indeed, "got hold of it" and the result is a new model actually reviving the reciprocating engine in cosmetics.

The rapid increase in the complexity has attracted the serious attention of engineers, both engine and aircraft, and strong efforts are now being made to not only stop this trend but to simplify the hardest engine installation problem as possible. But the major factors revealed in the problems offer seemingly small hope of success in these efforts.

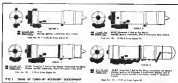
• **Aircraft Contribution**—Merely, this is correct in budget installation complexity is not all the fault of the engine. The airplane itself has increased enormously in complexity. The brief post war period being responsible for nearly as much increase as the preceding 20 yr.

For example, the generating capacity of the Douglas DC-6 is 31,000w. in contrast to 11,100w. in the DC-4 and 7800w. in the DC-3.

rates include submersibles, self-propelled, manned, unmanned, surface drive buoys, remote-sensing instruments, and child-defeating, power-operated trim tabs and a variety of fully-automatic devices, all of which were unknown to the fighter pilot of yore.

It was to insure that these power requirements, as well as those posed by maximum special areas and full-automatic controls that the taskset has increased in complexity. Thus, there is little question but what each additional security, device and control that has been added to it has been in response to a genuine need. But it is the character of the "need" that requires the larger situation.

How Equipment Grew:The "one moving part" takeoff of the early Whittle type was an excellent example of mechanical simplicity but, unfortunately, it was this very simplicity that created limitations on the engine's operation and performance.



One of the last steps was the addition of a float to control that automatically compensated for altitude changes as air inlet temperature and density. A turbine oil-cooled governor was added and it was only then necessary to install a fuel filter because of the screen effects of foreign matter on the burner.

At some experiments gone, a emergency fuel system was installed on all tankers. This installation was not strictly compensated since an "emergency" supply of fuel would only logically be necessary during takeoff, climb or in the landing approach. However, new engines must incorporate an automatically compensated emergency fuel system and, in addition, it must provide means for burning acceleration and deceleration to prevent "blowout" or excessive exhaust temperatures.

A line requirement, too, is an automatic shutoff device requiring no technique on the part of the pilot and which will prevent "hot starts" and their attendant hazards.

One of the early complaints added to the tailgate engine was an adjustable exhaust nozzle. Several designs have been developed experimentally. These consist of either a movable "ball" in the tailpipe or an adjustable nozzle area. Although the possibility of these mechanisms has not yet been proved (they were tried but abandoned in several jet fighters), there is not much question but what they will be required in the future when operating conditions become more variable.

► **Performance Degrade:** Increased performance demands on the engine have contributed substantially to its increased complexity. Since reduced fuel consumption, one of the major objectives, is obtained by increased compression and operating temperatures, these methods are being used progressively with attendant increase in weight.

Next major addition to the turbine will be turbine blade cooling, as even at its maximum intensity can be reduced sufficiently to permit peacked production.

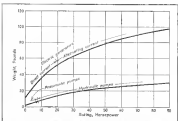


Fig. 2 Weight comparison of electric generation, hydraulic and pneumatic pumps

AVIATION WEEK, March 7, 1946

ENGINEERING

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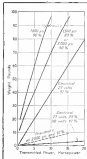


Fig 5: Graph of weight analysis of vacuum assist distribution systems. Comparison is made on a basis of delay in deliver power over a distance of 100 ft. at different power uses and efficiencies.

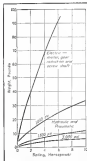


Fig 7: Graph for comparison of rising and weight at thrust solution for operating scenarios.

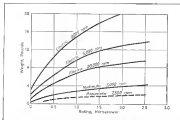


Fig 6: Rating and weight comparison of three types of constant duty rotary motors.

The greatly increased power provided by the turbopump over its predecessor piston engine has opened search to new regions of acceleration, altitude and maneuver that demand increased strength, engine longevity and increasing output and required fuel and lubricating systems. These high G loads and gyroscopic loads have forced designers to increase structural strength and to replace many assemblies for early mode of light alloys with high strength steel.

► **Two views**—One of the basic faults between engine and aircraft designer is the subject of economy and who is responsible for what cost.

The aircraft designer feels that if an economy is a problem for the engine designer, should the latter find that his job is simply to deliver a basic engine, his responsibility ending where the aircraft engineer's line attaches at the engine.

While this so-called "batter" has been merely business interplay between the two interests, it has also become a question of major importance and one not easily resolved by even the most informed engineer.

It is now no longer a question of whose responsibility a certain system is, but rather a question of how a particular service can be supplied with a minimum of weight and complexity.

► **Accessory Role**—Electricity is in the accessory that the complexity of the turbopump has assumed critical importance. Fig 1 illustrates this point clearly and indicates the rapid increase in size, weight and power requirements of accessories in turbopump installations.

Engine A represents a bare power package, suitable for use as a "booster" engine that is, first, the provision of thrust alone to the aircraft. The ac-

cessories installed are for the use of the engine alone and include only starting, ignition, lubrication, fuel and control systems. In this case the total weight of the accessories amounts to 11.4 percent of the engine weight.

Engine B begins the addition of aircraft accessories, that is, access of power required for the operation of equipment on the engine outputs to the engine itself.

In this engine there are added a generator and an accessory gear pad for the use of either a hydraulic pump or a vacuum pump according to the design of the turbine designer.

In addition, an emergency fuel system has been added to the engine itself, making a total accessory weight of 17.9 percent of the engine weight.

Engine C marks the use by the turbine engineer, at the source of high-pressure air sustained in the compressor casing of the turbopump, and as an air bleed is provided for this purpose. Also, a fuel drain is added to prevent accumulation of fuel within the engine. The output of the accessory drive has been increased 50 percent and all of these additions increase the weight of the accessories to 14.7 percent of engine weight.

► **Engine Under Development**—Engine D is a type now undergoing development but represents an existing requirement and typifies the present situation.

The aircraft accessory power has been increased another 11 percent and the engine features numerous additional items of equipment to serve requirements set forth by the customer for this case (the Navy Department).

This engine is equipped with its own air receiver, a fuel booster pump, warning equipment, a complete emergency master system and the addition of a variable area nozzle. (Continued p. 37)

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body and actuator units are individually engineered to meet your specific requirements. They are developed in a unit, made in the same plant, and assembled and tested together. No compromise need ever be made. You have one source of supply, one source of quality and one source of responsibility. You are saving, purchasing and assembling one item that is reliable in a minimum. Make this comparison and you'll see for yourself why WHITTAKER valves are first choice among the leading aircraft manufacturers the world over. Write WHITTAKER CO., Ltd., 313 N. GUYTON AVENUE, LOS ANGELES 36, CALIFORNIA.



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Ever since internal operations of aircraft mechanisms became inaccessable, Pesco has been specializing in building fuel and hydraulic equipment for America's commercial and military planes.

From the days of the first hydraulic pumps to today, when you still find Pesco equipment in 36 places on our largest jet liners, the design and construction of this equipment has been guided by the unvarying requirements of the aircraft industry. Because plane performance and human lives have been at stake, there has been no compromise with top quality and performance.

That's why Pesco developed "Fuselage Loading" for hydraulic pumps, an exclusive patented feature that automatically compensates for wind and maintains highest operating efficiency over a wide range of true pressure and altitude. That's why Pesco designs and builds its own electric motors — an all steel motor-drive unit as an integral part of the hydraulic or fuel unit, ensuring top operating efficiency. That's why, today, America's leading makers of jet engines have manufactured exclusively on Pesco high-pressure fuel pumps.

Just advantages of the engineering skill and "know-how" All the functions of Pesco — the largest manufacturers of specialized aircraft fuel and hydraulics in the world are at your service

KEY TO THE Pesco Precision Equipment Indicated above:

1. High-pressure hydraulic pump.
2. Hydraulic fueling pump.
3. Oil transfer pump.
4. Surface control booster pump.
5. Hydraulic motor drive hydraulic pump.
6. Hydraulic pressure relief valves.
7. Hydraulic flow regulator.
8. Pressure reducing valve.
9. Engine-driven fuel pump.
10. Motor drive fuel pump to pump.
11. Fuel transfer pump.
12. Engine-driven vacuum pump.
13. Oil separator.
14. Solenoid relief valve.
15. Air pressure relief valve.
16. Electric Motor for Cabin Ventilation.
17. Cabin heater fuel pump.

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All of this equipment adds up to 10 to 20 percent of the engine gross weight, or almost three times that of the original engine.

It is also important to retain the maximum engine frontal area created by the addition of these accessories, making impossible the "colder air effect" so long advanced as an engine attribute of the jet engine.

This increase adversely affects the "thrust per unit frontal area" often used as a basic criterion of the jet engine in comparison with other types.

► **Accessory Compromise**—One of the approaches to a solution to this problem is a study of various types of accessories and the time and weight of their operating cycles. In a comparison between electrical, hydraulic and pneumatic operating equipment, there is no question but what the pneumatic pump has the lightest weight per horsepower delivered.

This comparison is shown in Fig. 1, which also indicates that although the pneumatic pump is extremely efficient at comparatively low power outputs, there are not available for comparison pneumatic pumps at power ratings above about 5 hp.

Next most efficient, as a hp/tp basis, is the hydraulic pump, which is 34 times as light as the electric gas turbine throughout its range of application.

Most expensive in weight is the air pressure system, it requires a constant speed drive and heavy construction in its own right.

From the point of view of the engine alone, the pneumatic pump is the most efficient means of providing power for the airplane but a consideration of the pneumatic system alone does not provide a solution to the problem, for after leaving the engine the aircraft designer must consider the efficiency of utilization of this power within the airplane.

► **Combination Factor**—Fig. 1 shows a comparison of the weight and power of various distribution systems for aircraft and here the picture is changed in a way.

A high voltage electrical wiring system will transmit the same power at about 1/4 the weight of a pneumatic system. This advantage is lost, however, when lower voltages are used.

Electrical wiring equipment at the total weight of an electrical system has the considerable advantage, however, in its simplicity and flexibility of installation.

After studying and distributing this energy, the designer has next the problem of efficient utilization, and Figs. 4 and 5 examine the weight per horsepower of wiring and versus utilization.

In Fig. 4, type of conductor duty

rating ratios are compared, with the pneumatic type showing the highest efficiency. The power efficiency is calculated by the volume of air required with the efficiency depending on the speed is concerned.

Fig. 5 compares the volume on the basis of there are in lower altitudes and how the hydraulic and pneumatic systems are practically obtained and as all cases for superior to the electric system. The latter, in order to convert its energy into a mechanical form action around a motor, a reducing gear and a jack, whereas the hydraulic and pneumatic systems need only a cylinder containing a piston.

► **Comparison**—Even these comparisons are not complete, for an accurate analysis of the various systems involved.

For example, the hydraulic system requires a motor, the pneumatic system requires a compressor at altitude due to the low air density, the electrical system (d.c.) requires a converter in the form of a battery.

For these reasons, engineers are going to the electric system for their search for accessory system efficiency.

Examples of these manufacturers include the use of electrically operated, accurately controlled selector valves in a hydraulic system, hydraulic pumps driven by small, high speed electric motors and an electric motor-driven air compressor for pneumatic boosters have also operated.

Even detailed comparisons of the various systems, in a particular installation, as the selection of an air as the most efficient, there now appear to be doubt that the electric system will be required in the future.

Thus, electrical energy is necessary for the operation of take-off systems, many instruments and for control systems, whether it is an efficient method of power utilization or not.

Availability of high pressure air in the compressor casing may eliminate the need for a pressure pump on the aircraft and give a high pressure air for cabin air conditioning, pressurization and the operation of seats in which speed is important.

The hydraulic jet reaction as an extremely efficient method of converting large amounts of energy, such as it is required for heavy bomber landing gear operation. Its pressure-equalizing characteristic also makes it ideal for control boost systems, automatic gear controls and similar uses in control units.

► **Auxiliary Power**—Most engineers advise the auxiliary power unit is the most desirable addition to their power plant. It would remove all accessories from the engine and eliminate the losses

power drain that directly affects the performance of the engine.

However, size and weight of auxiliary power units present serious problems, single-engine craft. An alternative solution is the use of an accessory drive gear box remote from the engine and connected by a single shaft to a single power take-off on the engine.

This would simplify the design problem of both the engine and aircraft construction, since the engine designer could be relieved of the added weight, limited size and complexity of the auxiliary accessories and the aircraft designer could simplify the installation and maintenance of the powerplant while placing the accessories in a more desirable location, remote both from the engine and an accessibly viewpoint.

Solution is still sought for the effects of increased high altitude on accessory systems, a problem in which there is little available data.

Sometimes, spend may also create additional problems in manufacturing loads and stress effects on both wings and fuselage.

Obviously, aerodynamic engineers have interfused and multiplied the security problems of aircraft and engine designers and certain needs for additional materials and development work will be needed the steadily heavier loads of providing powerful efficient turbopropellers for our high performance aircraft.

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Fuselage Production Up

Yonkers Products, Inc., produced more than one and one-quarter billion of its speed units, clips and clamps during 1946. For the first time in its history Yonkers produced more than 100,000 speed units a month. Yonkers is building new production facilities to add a speed unit clip station to its line. The clip is used of eliminating the need for stitching in many assembly techniques.

EDITORIAL

No Gains Without Safety

The Flight Safety Foundation is unique in aviation. Nothing like this nonprofit, independent educational group has existed before. In a field of transportation that has suffered so much from accident information or successfully sensational publicity, the Foundation has a tremendous task.

To date, financial aid has been disappointing. The Foundation's fund raising program is continuing with little fanfare or promotion, but the goal for 1979 is at least \$100,000.

An official summary of the accomplishments of the Foundation in its first five months of active life is on page 30. This flight is made available by the Foundation's director, Jerome F. Lederer, from the headquarters at 515 Madison Ave., in New York City.

1. It has established an experience interchange service which provides the aviation industry with a channel for exchanging data on accidents, near-accidents, missing procedures and related safety information. The improved operational practices developed by one aircraft operator are thereby made known to other operators and to the aircraft manufacturers. This information exchange also has opened dissemination of accident prevention information within the individual organizations.

2. It initiated a forum where aeronautical engineers and aircraft operators met in a two day conference to exchange ideas on internal company policy as related to safety. Representatives of freight and domestic airlines as well as aircraft manufacturers and the military services participated in the forum. The results were fruitful, especially in relating methods for safety education and analysis of operations from the viewpoint of safety.

3. Accurate reports on accidents are necessary to provide educational material and prevent recurrence. Re sponsibility for investigation of rising aircraft accidents is being delegated by the CAB to state police or state aviation agencies. The Foundation requested and conducted a seminar for the training of civilian aircraft accident investigators, the first of its kind in the country and probably in the world. This was accomplished with the assistance of experts furnished by CAB and CAA. Lieut. Gen. George E. Straszmeier cooperated in providing the facilities for the seminar at Mitchell Air Force Base.

4. The Foundation has advised the Coast Guard in advising transoceanic operation of the search and rescue organization at hand when a ditching occurs at sea and has published the procedures recommended for use by captains of a merchant vessel when an airplane prepares to ditch in his vicinity.

5. After learning that one airline was using Coast Guard pilots to train its pilots in ditching, the Founda-

tion speeded the use of Coast Guard pilots to train other airline pilots in ditching procedures.

6. The Foundation has produced the first integrated study of scientific methods for evaluating flight crew assignments for large transport aircraft. The project, an outgrowth of a great deal of prior study, was sponsored by certain airlines and carried out in cooperation with other interested organizations.

7. The Foundation has signed a contract with the CAA to develop this study further.

8. It is working with other organizations on development of safety programs applying particularly to private flying. Like the Automotive Safety Foundation, the Flight Safety Foundation will work with whatever organization is best suited to accomplish a particular project.

9. It has established a research project to study cockpit simplification and reduce causes of fatal accidents in small private aircraft. This project was financed in memory of Richard B. Depey, a leading exponent of safety in private flying. A preliminary report on this project will be available soon.

10. The Foundation is constantly analyzing accident data to show where safety efforts are likely to produce maximum results and it helps other organizations analyze their safety projects so that practical safety measures will not be allowed to migrate, but rather be propelled into adoption.

This is an impressive record for less than a year's operation.

Already, a number of public spirited individuals and companies have contributed to this nonprofit, impartial group dedicated to the advancement of knowledge for safety in flight.

Contributors to date include three leaders Alexander & Alexander (Insurance), American Airlines, Beech Aircraft, Boeing Aircraft, Braniff Airways, William A. M. Burdick, Capital Airlines, Colonial Airlines, Connecticut General Life Insurance Co., Continental Air Lines, Delta Air Lines, Sherran Farfield, The Farmship Fund, Mrs. L. Garrett Foss, Grumman Aircraft Engineering Corp., Los Angeles Airways, National Aeronautics Corp., Edward J. Noble, Ohio Oil Co., Pan American-Grace Airways, Lawrence S. Rockefeller, Sportmen Pilots Assn., United Air Lines, Standard Oil Co. of California.

These contributions are worthy of commendation. It is regrettable that their number is so small. At this date the goal for 1979 appears distant, and unattainable. But we have more help in aviation's burden than to believe the Foundation will fail to win its rightful support from those it will serve best. In aviation there are no good without safety.

ROBERT H. WOOD

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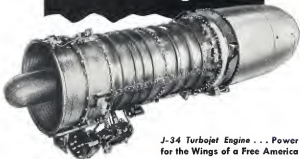
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